

Research Article

Effect of Rotation on Wave Propagation in Micro-elongated Thermo-elastic Media with Fractional Conformable Derivative under the Refined Dual-Phase-Lag Model

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Abstract:

This study investigates the impact of rotation on wave propagation within a micro-elongated thermo-elastic medium, employing the fractional conformable derivative under Lord-Shulman (L-S) theory, the Dual-Phase-Lag (DPL) model, and the Refined Dual-Phase-Lag (RDPL) model. The governing equations for heat conduction, mechanical motion, and micro-elongation are formulated to account for finite thermal wave speeds and microstructural effects.

By applying non-dimensionalization and normal mode analysis, the coupled system is transformed into analytically solvable form. Explicit solutions for displacement, temperature, stress, and micro-elongation fields are obtained. Numerical results compare L-S, DPL, and RDPL models with and without rotation, and assess the effect of fractional order.

The results show that rotation significantly influences wave propagation characteristics such as amplitude, speed, and attenuation.

Keywords: Micro-elongated; Refined Dual-Phase-Lag Model; Rotation; Fractional Conformable Derivative

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List of Nomenclature

Quantity	Definition
\mathbf{u}	Displacement vector,
$\alpha_{t_1}, \alpha_{t_2}$	Coefficient of linear thermal expansion, where $\gamma_0 = (3\lambda + 3\mu) \alpha_{t_1}$, $\gamma_1 = (3\lambda + 3\mu) \alpha_{t_2}$,

Quantity	Definition
λ and μ	Constants of Lamé,
j^*	Microinertia,
T_0	Reference temperature,
Ω	Initial stress,
$\lambda_0, \lambda_1, a_0$	Constants of micro-elongational,

Quantity	Definition
ψ	Micro-elongational scalar,
Θ	Temperature,
c_e	Specific heat,
ρ	Density,
t	Time,
k	Thermal conductivity,
τ_θ	Phase-lag of the temperature gradient,
σ_{ij}	Component of stress tensor,
τ_q	Phase-lag of the heat flux.

1. Introduction

Micro-elongated thermo-elastic materials are part of a wider category of generalized continua that extend beyond the conventional deformation behavior of solids by incorporating additional micro-elongation affects. Unlike classical thermo-elastic theories, which describe mechanical behavior solely through displacement fields and thermal behavior via temperature variations, micro-elongated theories introduce an extra kinematic variable representing internal micro-level stretching. This internal elongation can correspond to phenomena such as fiber extension, molecular chain deformation, pore-structure distortion, or reorientation of microstructural elements within the material. By including this micro-elongation field, the thermo-mechanical response of complex materials such as composite media, biological tissues, polymeric substances, porous solids, and engineered microstructured materials can be described more comprehensively, as internal components are allowed to deform independently of the overall bulk motion. When coupled with thermal affects, micro-elongation interacts with the temperature field, generating additional stress contributions and heat transfer mechanisms that are absent in classical thermo-elastic models. Consequently, wave propagation, stress development, and temperature evolution in micro-elongated media exhibit distinctive features including attenuation, dispersion, and relaxation affects governed by the underlying microstructure (see [1, 2, 3, 4, 5, 6, 7, 8]). This theoretical approach is particularly valuable for analyzing dynamic phenomena, thermo-elastic interactions, and materials subjected to thermal loading, where delayed and spatially varying internal responses cannot be accurately captured by traditional thermo-elastic frameworks.

Fractional thermo-elasticity is an advanced generalization of classical thermo-elastic theories, incorporating fractional-order derivatives into the equation of energy. Within this framework, heat conduction is extended using fractional derivatives, allowing the model to capture non-Fourier heat transfer phenomena, such as finite-speed thermal wave propagation and dispersive affects. The inclusion of fractional parameters enhances the model's flexibility, improving the description of thermal lag, damping, wave attenuation, and dynamic stability in complex materials. As a result, fractional thermo-elasticity provides a more accurate and adaptable approach for analyzing modern engineering systems and advanced material structures, effectively rep-

resenting behaviors that classical theories cannot fully capture. For further foundational studies on this subject, readers are referred to [9, 10, 11, 12, 13, 14, 15, 16, 17, 18, 19, 20].

The RDPL model introduces a sophisticated extension of heat conduction theories and serves as an improved version of the original DPL model. While the DPL model accounts for two distinct phase lags one associated with the temperature gradient and the other with the heat flux, it often falls short in capturing the complex thermal delay mechanisms that occur in real materials, particularly when they are subjected to rapid or high-intensity thermal excitations. These shortcomings become especially pronounced in scenarios involving ultrafast thermal processes, such as laser-induced heating, advanced micro-electronic cooling technologies, and the propagation of thermo-elastic waves, where heat transfer deviates significantly from the classical diffusive behavior. To overcome these limitations, the RDPL model represents more advanced features, including higher-order terms and carefully adjusted phase-lag parameters, which together provide a more refined representation of the thermal response. This enhanced structure allows the model to more accurately reflect the finite speed at which thermal disturbances propagate and to better represent the intrinsic properties of materials. Consequently, the RDPL model effectively describes thermal transport in regimes where heat does not spread instantaneously, but travels with a measurable delay, exhibiting characteristics similar to those of thermal waves. In addition to its improved physical accuracy, Comprehensive explanations, derivations, and theoretical insights into the RDPL model are available in references [21, 22, 23, 24, 25, 26, 27, 28, 29]. Because of its increased precision and stability, the RDPL model has become an important tool for analyzing thermo-elastic behavior in applications.

The influence of rotation on thermo-elasticity plays a crucial role in accurately describing the mechanical and thermal responses of solids subjected to high-speed or angular motion. When a thermo-elastic medium undergoes rotation, additional inertial forces, namely the Coriolis force, centrifugal force, and Euler acceleration become significant and modify the governing field equations. These rotational affects introduce coupling among the elastic and thermal fields and lead to substantial changes in wave propagation characteristics, stress and temperature. In rotating systems, the centrifugal force tends to stretch or compress the material depending on the angular velocity and radial position, resulting in modified equilibrium states and altered deformation patterns. Meanwhile, the Coriolis force influences the propagation of thermo-elastic waves, producing directional asymmetry and frequency shifts that are absent in non-rotating media. Furthermore, rotation can enhance or suppress thermal diffusion depending on the material constants and rotational speed, leading to non-uniform temperature fields and dynamic instabilities under certain conditions for more details, see [30, 31, 32,

33, 34, 35, 36, 37, 38]. Such affects become especially important in advanced engineering applications, including rotating machinery, spacecraft structures, turbine blades, micro-electromechanical devices, and geophysical models of the Earth’s interior, where both rapid rotation and thermal loading coexist. Incorporating rotation into thermo-elastic theories allows for more realistic predictions of thermal stresses, wave speeds, damping behavior, and energy transport, ensuring the reliability and safety of structures operating in rotating environments.

Although thermo-elasticity has been widely studied using phase-lag models and fractional derivatives, and the mechanical behavior of micro-elongated materials has been investigated separately, a significant gap remains in the literature regarding their combined interaction within the fractional-order RDPL model. Most existing studies treat these affects independently, focusing either on fractional-order deformation or on non-Fourier heat conduction, while neglecting the simultaneous influence of micro-elongation and rotation. To date, no reported work has unified micro-elongated behavior, rotational affects, fractional derivatives, and the RDPL model within a single thermo-elastic formulation. Therefore, a primary objective of the present study is to fill this gap by developing a comprehensive model that captures the coupled thermo-mechanical and fractional-order responses of rotating micro-elongated materials under the RDPL model.

This paper investigates how rotation and fractional conformable derivatives affect the thermo-elastic behavior of a micro-elongated half-space within L-S theory, DPL model, and RDPL model. The governing equations for motion, heat conduction, and micro-elongation are first established and then rewritten in dimensionless form. Using Normal Mode Analysis, the system of partial differential equations is reduced to ordinary differential equations, and the boundary conditions at $z = 0$ are applied to determine the necessary solution constants. After obtaining the analytical solutions, numerical and graphical analyses are performed to study the affects of rotation and fractional parameters on displacement, stress fields, and micro-elongation. The results clearly show that both rotation and fractional-order affects significantly modify the thermo-micro-elongated response of the medium under the three considered models.

2. Governing equations

This section presents the principal governing equations that characterize the 2–dimensional micro-elongated thermo-elastic response as formulated within the frameworks of the L-S theory, the DPL model, and the RDPL model. Particular emphasis is placed on examining how rotational affects the behavior of the system. For the sake of mathematical convenience and to enable more effective analytical treatment, the governing relations are subsequently transformed into a nondimensional form.

The equation of motion can be expressed in the fol-

lowing form [39, 40]

$$\sigma_{ij,j} = \rho [u_{i,tt} + \{\boldsymbol{\Omega} \wedge (\boldsymbol{\Omega} \wedge \mathbf{u})\}_i + (2\boldsymbol{\Omega} \wedge \mathbf{u}, t)_i], \quad (1)$$

where

$$\sigma_{ij} = 2\mu\varepsilon_{ij} + (\lambda e - \gamma_0\Theta + \lambda_0\psi) \delta_{ij}. \quad (2)$$

The equation of micro-elongation can be displayed in the following form [41]

$$a_0\psi_{,ii} + \gamma_1\Theta - \lambda_1\psi - \lambda_0u_{j,j} = \frac{\rho j^* \psi_{,tt}}{2}. \quad (3)$$

The equation of energy can be written in the following form [23, 42]:

$$\begin{aligned} 0 = & k \left(1 + \sum_{n=1}^N \frac{\tau_\theta^{n-1+\alpha}}{\Gamma(n+\alpha)} \frac{\partial^\alpha}{\partial t^\alpha} \frac{\partial^{n-1}}{\partial t^{n-1}} \right) \Theta_{,ii} \quad (4) \\ & - \gamma_1 T_0 \psi_{,t} - \left(1 + \sum_{n=1}^N \frac{\tau_q^{n-1+\alpha}}{\Gamma(n+\alpha)} \frac{\partial^\alpha}{\partial t^\alpha} \frac{\partial^{n-1}}{\partial t^{n-1}} \right) \\ & \times [\gamma_0 T_0 (u_{,xt} + w_{,zt}) + \rho c_e \Theta_{,t}], \end{aligned}$$

where, $\frac{\partial^\alpha}{\partial t^\alpha}$ is the conformable fractional derivative of order $0 < \alpha \leq 1$ [43].

When

- 1- $N = 3$ and $(\tau_\theta < \tau_q \neq 0, k \neq 0)$ is RDPL model,
- 2- $N = 1$ and $(\tau_\theta < \tau_q \neq 0, k \neq 0)$ is DPL model,
- 3- $N = 1$ and $(\tau_\theta = 0, \tau_q \neq 0, k \neq 0)$ is L-S theory.

Definition [44, 45]. Let f be a function such that $f : [a, \infty) \rightarrow \mathbb{R}$. Then, the conformable derivative of a function f is defined as

$$D_t^\alpha f(t) = \lim_{\epsilon \rightarrow 0} \frac{f(t + \epsilon t^{1-\alpha}) - f(t)}{\epsilon},$$

for all $t \geq a, \alpha \in (0, 1]$.

In the RDPL model formulation, setting $N=3$ introduces higher-order phase-lag terms that extend the classical DPL model and allow a more accurate description of heat conduction in microstructured materials. In Equation (4), the coupling term $(-\gamma_1 T_0 \psi_{,t})$ represents the direct interaction between the thermal field and the micro-elongation variable (ψ), indicating that the rate of micro-elongation contributes to the temperature evolution. The inclusion of higher-order terms enhances the model’s capability to capture thermo–microstructural interactions and finite thermal wave propagation.

From Eq. (2) in Eq. (1) with the aid of the displacement vector $\mathbf{u}(x, z, t) = u(u, 0, w)$ and rotation $\boldsymbol{\Omega}$, we see

$$\begin{aligned} \mu \nabla^2 u + (\lambda + \mu) e_{,x} - \gamma_0 \Theta_{,x} + \lambda_0 \psi_{,x} \quad (5) \\ = \rho [u_{,tt} - \Omega^2 u + 2\Omega w_{,t}], \end{aligned}$$

$$\begin{aligned} \mu \nabla^2 w + (\lambda + \mu) e_{,z} - \gamma_0 \Theta_{,z} + \lambda_0 \psi_{,z} \quad (6) \\ = \rho [w_{,tt} - \Omega^2 w - 2\Omega u_{,t}]. \end{aligned}$$

Employing the displacement vector $\mathbf{u}(x, z, t) = u(u, 0, w)$ in Eq. (3), we infer

$$a_0\psi_{,ii} + \gamma_1\Theta - \lambda_1\psi - \lambda_0(u_{,x} + w_{,z}) = \frac{\rho J^* \psi_{,tt}}{2}. \quad (7)$$

By applying the properties of the conformable derivative as outlined in [46] to Eq. (4), the equation takes the following form:

$$0 = k \left(1 + \sum_{n=1}^N \frac{\tau_\theta^{n-1+\alpha}}{\Gamma(n+\alpha)} (t^{1-\alpha}) \frac{\partial^n}{\partial t^n} \right) \Theta_{,ii} \quad (8)$$

$$- \gamma_1 T_0 \psi_{,t} - \left(1 + \sum_{n=1}^N \frac{\tau_q^{n-1+\alpha}}{\Gamma(n+\alpha)} (t^{1-\alpha}) \frac{\partial^n}{\partial t^n} \right) \times [\rho c_e \Theta_{,t} + \gamma_0 T_0 (u_{,xt} + w_{,zt})].$$

In this analysis, we employ the following nondimensional variables to streamline the governing equations

$$(x, z) = \frac{c_1}{\omega^*} (\bar{x}, \bar{z}), \quad (9)$$

$$(u, v) = \frac{T_0 \gamma_0}{\rho c_1 \omega^*} (\bar{u}, \bar{v}),$$

$$(t, \tau_\theta, \tau_q) = \frac{(\bar{t}, \bar{\tau}_\theta, \bar{\tau}_q)}{\omega^*},$$

$$\Theta = T_0 \bar{\Theta}, \quad \sigma_{ij} = \gamma_1 T_0 \bar{\sigma}_{ij},$$

$$\psi = \frac{\gamma_0 T_0}{\lambda_0} \bar{\psi}, \quad \Omega = \omega^* \bar{\Omega},$$

$$k\omega^* = \rho c_1^2 c_e, \quad \rho c_1^2 = \lambda + \mu.$$

Utilizing Eq. (9) in Eqs. (5), (6), (7), and (8), one uncovers

$$\left(a_1 \nabla^2 - \frac{\partial^2}{\partial t^2} + a_2 \frac{\partial^2}{\partial x^2} + \Omega^2 \right) u + \left(a_2 \frac{\partial^2}{\partial x \partial z} - 2\Omega \frac{\partial}{\partial t} \right) w - \frac{\partial \Theta}{\partial x} + \frac{\partial \psi}{\partial x} = 0, \quad (10)$$

$$\left(a_2 \frac{\partial^2}{\partial x \partial z} + 2\Omega \frac{\partial}{\partial t} \right) u + \left(a_1 \nabla^2 - \frac{\partial^2}{\partial t^2} + a_2 \frac{\partial^2}{\partial z^2} + \Omega^2 \right) w - \frac{\partial \Theta}{\partial z} + \frac{\partial \psi}{\partial z} = 0, \quad (11)$$

$$a_5 \frac{\partial u}{\partial x} + a_5 \frac{\partial w}{\partial z} - a_3 \Theta - \left(\nabla^2 - a_4 - a_6 \frac{\partial^2}{\partial t^2} \right) \psi = 0, \quad (12)$$

$$a_8 (1 + \tau_2) \frac{\partial^2 u}{\partial x \partial t} + a_8 (1 + \tau_2) \frac{\partial^2 w}{\partial z \partial t} - \left(\nabla^2 + \tau_1 \nabla^2 - a_7 \frac{\partial}{\partial t} - \tau_2 a_7 \frac{\partial}{\partial t} \right) \Theta + a_9 \frac{\partial \psi}{\partial t} = 0, \quad (13)$$

where, $a_1 = \frac{\mu}{c_1^2 \rho}$, $a_2 = \frac{\mu + \lambda}{c_1^2 \rho}$, $a_3 = \frac{\lambda_0 \gamma_1 c_1^2}{\gamma_0 a_0 \omega^{*2}}$, $a_4 = \frac{c_1^2 \lambda_1}{a_0 \omega^{*2}}$, $a_5 = \frac{\lambda_0^2}{a_0 \rho \omega^{*2}}$, $a_6 = \frac{\rho J^* c_1^2}{2a_0}$, $a_7 = \frac{\rho c_e c_1^2}{k \omega^*}$, $a_8 = \frac{\gamma_0^2 T_0}{k \omega^* \rho}$, $a_9 = \frac{\gamma_0 \gamma_1 T_0 c_1^2}{k \omega^* \lambda_0}$, $\tau_1 = \sum_{n=1}^N \frac{i \omega \tau_\theta^{n-1+\alpha}}{\Gamma(n+\alpha)} (t^{1-\alpha}) \frac{\partial^n}{\partial t^n}$, and $\tau_2 = \sum_{n=1}^N \frac{i \omega \tau_q^{n-1+\alpha}}{\Gamma(n+\alpha)} (t^{1-\alpha}) \frac{\partial^n}{\partial t^n}$.

Employing Eq. (9) in Eq. (2), we obtain

$$\sigma_{xx} = \frac{\partial u}{\partial x} + a_{10} \frac{\partial w}{\partial z} - \Theta + \psi, \quad (14)$$

$$\sigma_{zz} = a_{10} \frac{\partial u}{\partial x} + \frac{\partial w}{\partial z} - \Theta + \psi, \quad (15)$$

$$\sigma_{zx} = \sigma_{xz} = a_1 \frac{\partial u}{\partial z} + a_1 \frac{\partial w}{\partial x}. \quad (16)$$

where, $a_{10} = \frac{\lambda_0^2}{a_0 \rho \omega^{*2}}$.

The displacement potentials $\eta(x, z, t)$ and $\xi(x, z, t)$ which relate to displacement components have been introduced, we realize

$$u = \eta_{,x} + \xi_{,z}, \quad w = \eta_{,z} - \xi_{,x}. \quad (17)$$

Using Eq. (17) in Eqs. (10), (11), (12), and (13), one obtains

$$\left[(a_1 + a_2) \nabla^2 + \Omega^2 - \frac{\partial^2}{\partial t^2} \right] \eta + 2\Omega \xi_{,t} - \Theta + \psi = 0, \quad (18)$$

$$2\Omega \eta_{,t} - \left[a_1 \nabla^2 + \Omega^2 - \frac{\partial^2}{\partial t^2} \right] \xi = 0, \quad (19)$$

$$a_5 \nabla^2 \eta - a_3 \Theta - \left(\nabla^2 - a_4 - a_6 \frac{\partial^2}{\partial t^2} \right) \psi = 0, \quad (20)$$

$$a_8 (1 + \tau_2) \nabla^2 \left(\frac{\partial \eta}{\partial t} \right) \quad (21)$$

$$- \left(\nabla^2 + \tau_1 \nabla^2 - a_7 \frac{\partial}{\partial t} - \tau_2 a_7 \frac{\partial}{\partial t} \right) \Theta + a_9 \frac{\partial \psi}{\partial t} = 0.$$

3. Solution

The solution of the physical field variables can be represented and examined using normal mode analysis as follows:

$$[u, w, \Theta, \psi, \eta, \xi, \sigma_{ij}] (x, z, t) \quad (22)$$

$$= [\tilde{u}, \tilde{w}, \tilde{\Theta}, \tilde{\psi}, \tilde{\eta}, \tilde{\xi}, \tilde{\sigma}_{ij}] (z) e^{i(-bx + \frac{\omega t \alpha}{\alpha})}.$$

where, ω is the wave number and b is the frequency

Inserting Eq. (22) into Eqs. (10), (11), (12), and (13), we realize

$$\left(\varsigma_1 D^2 + \varsigma_2 \right) \tilde{\eta} + \varsigma_3 \tilde{\xi} - \tilde{\Theta} + \tilde{\psi} = 0, \quad (23)$$

$$\varsigma_3 \tilde{\eta} - \left[a_1 D^2 + \varsigma_4 \right] \tilde{\xi} = 0, \quad (24)$$

$$\left[a_5 D^2 - \varsigma_5 \right] \tilde{\eta} - a_3 \tilde{\Theta} - \left(D^2 + \varsigma_6 \right) \tilde{\psi} = 0, \quad (25)$$

$$\left[\varsigma_7 D^2 - \varsigma_8 \right] \tilde{\eta} - \left(\varsigma_9 D^2 - \varsigma_{10} \right) \tilde{\Theta} + \varsigma_{11} \tilde{\psi} = 0, \quad (26)$$

where, $\varsigma_1 = a_1 + a_2$, $\varsigma_2 = \omega^2 t^{2\alpha-2} - i(\alpha-1)\omega t^{\alpha-2} + \Omega^2 - b^2 \varsigma_1$, $\varsigma_3 = 2i\Omega \omega t^{\alpha-1}$, $\varsigma_4 = \omega^2 t^{2\alpha-2} - i(\alpha-1)\omega t^{\alpha-2} + \Omega^2 - a_1 b^2$, $\varsigma_5 = b^2 a_5$, $\varsigma_6 = a_6 [\omega^2 t^{2\alpha-2} - i(\alpha-1)\omega t^{\alpha-2}] - a_4 - b^2$, $\varsigma_7 = ia_8 (1 + \tau_2) \omega t^{\alpha-1}$, $\varsigma_8 = \varsigma_7 b^2$, $\varsigma_9 = (1 + \tau_1)$, $\varsigma_{10} = \varsigma_9 b^2 + i(1 + \tau_2) a_7 \omega t^{\alpha-1}$, $\varsigma_{11} = ia_9 \omega t^{\alpha-1}$.

It is observed that the cases

1. $\tau_1 = \frac{i\omega\tau_\theta^\alpha}{\Gamma(1+\alpha)} + \frac{\omega\tau_\theta^{1+\alpha}}{\Gamma(2+\alpha)} [it^{-1} - \omega t^{\alpha-1}] + \frac{\omega\tau_\theta^{2+\alpha}}{\Gamma(3+\alpha)} [i(\alpha-1)(\alpha-2)t^{-2} - 3\omega(\alpha-1)t^{\alpha-2} - i\omega^2 t^{2\alpha-2}]$ and $\tau_2 = \frac{i\omega\tau_q^\alpha}{\Gamma(1+\alpha)} + \frac{\omega\tau_q^{1+\alpha}}{\Gamma(2+\alpha)} [it^{-1} - \omega t^{\alpha-1}] + \frac{\omega\tau_q^{2+\alpha}}{\Gamma(3+\alpha)} [i(\alpha-1)(\alpha-2)t^{-2} - 3\omega(\alpha-1)t^{\alpha-2} - i\omega^2 t^{2\alpha-2}]$ correspond to the RDPL model
2. $\tau_1 = \frac{i\omega\tau_\theta^\alpha}{\Gamma(\alpha+1)}$ and $\tau_2 = \frac{i\omega\tau_q^\alpha}{\Gamma(\alpha+1)}$ correspond to the DPL model
3. $\tau_1 = 0$ and $\tau_2 = \frac{i\omega\tau_q^\alpha}{\Gamma(\alpha+1)}$ correspond to the L-S theory

Equations (23), (25), and (26) yield a nontrivial solution when the determinant of the coefficient matrix associated with the physical quantities equals zero. Using the MATLAB program, one acquires

$$\left(D^8 - A_1 D^6 + A_2 D^4 - A_3 D^2 + A_4 \right) \{ \tilde{\eta}(z), \tilde{\xi}(z), \tilde{\Theta}(z), \tilde{\psi}(z) \} = 0, \tag{27}$$

where, $A_1 = \frac{-1}{s_1 s_9 a_1} \{ -s_7 a_1 + s_9 a_1 a_5 + s_1 s_4 s_9 - s_1 s_{10} a_1 + s_2 s_9 a_1 + s_1 s_6 s_9 a_1 \}$,
 $A_2 = \frac{1}{s_1 s_9 a_1} \{ s_4 s_7 + s_3^2 s_9 - s_7 a_1 a_3 - s_{10} a_1 a_5 - s_{11} a_1 a_5 - s_1 s_4 s_{10} + s_2 s_4 s_9 - s_2 s_{10} a_1 - s_6 s_7 a_1 - s_5 s_9 a_1 + s_4 s_9 a_5 + s_1 s_4 s_6 s_9 - s_1 s_6 s_{10} a_1 + s_2 s_6 s_9 a_1 + s_1 s_{11} a_1 a_3 + s_8 a_1 \}$,
 $A_3 = \frac{-1}{s_1 s_9 a_1} \{ s_4 s_8 + s_3^2 s_6 s_9 - s_2 s_4 s_{10} - s_4 s_6 s_7 - s_4 s_5 s_9 - s_4 s_7 a_3 + s_6 s_8 a_1 + s_5 s_{10} a_1 + s_5 s_{11} a_1 - s_4 s_{10} a_5 - s_4 s_{11} a_5 - s_1 s_4 s_6 s_{10} + s_2 s_4 s_6 s_9 + s_1 s_4 s_{11} a_3 - s_2 s_6 s_{10} a_1 + s_2 s_{11} a_1 a_3 - s_3^2 s_{10} + s_8 a_1 a_3 \}$,
 $A_4 = \frac{1}{s_1 s_9 a_1} \{ s_4 s_6 s_8 + s_4 s_5 s_{10} + s_4 s_5 s_{11} + s_4 s_8 a_3 - s_3^2 s_6 s_{10} + s_3^2 s_{11} a_3 - s_2 s_4 s_6 s_{10} + s_2 s_4 s_{11} a_3 \}$. This equation can be expressed as

$$\left(D^2 - r_1 \right) \left(D^2 - r_2 \right) \left(D^2 - r_3 \right) \left(D^2 - r_4 \right) \{ \tilde{\eta}(z), \tilde{\xi}(z), \tilde{\Theta}(z), \tilde{\psi}(z) \} = 0, \tag{28}$$

where, r_1, r_2, r_3 and r_4 are roots

The general solution of Eq. (28) bound as $z \rightarrow \infty$ can be written as

$$\tilde{\eta}(z) = \sum_{m=1}^4 G_m e^{-r_m z}, \tag{29}$$

$$\tilde{\xi}(z) = \sum_{m=1}^4 \vartheta_{1m} G_m e^{-r_m z} \tag{30}$$

$$\tilde{\Theta}(z) = \sum_{m=1}^4 \vartheta_{2m} G_m e^{-r_m z}, \tag{31}$$

$$\tilde{\psi}(z) = \sum_{m=1}^4 \vartheta_{3m} G_m e^{-r_m z}. \tag{32}$$

where, $\vartheta_{1m} = \frac{s_3}{a_1 r_m^2 + s_4}$, $\vartheta_{2m} = \frac{s_7 r_m^4 + (s_6 s_7 - s_8) r_m^2 - s_5 s_{11} - s_6 s_8}{s_9 r_m^4 + (s_6 s_9 - s_{10}) r_m^2 - s_6 s_{10} - a_3 s_{11}}$, and $\vartheta_{3m} = - (s_1 r_m^2 + s_2) - s_3 \vartheta_{1m} + \vartheta_{2m}$.

Using Eq. (22) in Eq. (17), then employing Eqs. (29) and (30), one sees

$$\tilde{u} = -\sum_{m=1}^4 (r_m \vartheta_{1m} + ib) G_m e^{-r_m z}, \tag{33}$$

$$\tilde{w} = \sum_{m=1}^4 (ib \vartheta_{1m} - r_m) G_m e^{-r_m z}. \tag{34}$$

Utilizing Eq. (22) into Eqs. (14), (15), and (14), then using Eqs. (31), (32), (33) and (34), we infer

$$\tilde{\sigma}_{xx} = \sum_{m=1}^4 \vartheta_{4m} G_m e^{-r_m z}, \tag{35}$$

$$\tilde{\sigma}_{zz} = \sum_{m=1}^4 \vartheta_{5m} G_m e^{-r_m z}, \tag{36}$$

$$\tilde{\sigma}_{xz} = \tilde{\sigma}_{zx} = \sum_{m=1}^4 \vartheta_{6m} G_m e^{-r_m z}, \tag{37}$$

where, $\vartheta_{4m} = ib(1 - a_{10}) r_m \vartheta_{1m} + a_{10} r_m^2 - b^2 - \vartheta_{2m} + \vartheta_{3m}$, $\vartheta_{5m} = ib(a_{10} - 1) r_m \vartheta_{1m} + r_m^2 - a_{10} b^2 - \vartheta_{2m} + \vartheta_{3m}$, $\vartheta_{6m} = a_1 (r_m^2 + b^2) \vartheta_{1m} + 2ib a_1 r_m$.

4. Boundary condition

To find the constants (G_1, G_2, G_3 , and G_4), we have applied the boundary conditions for the problem at $z = 0$ as follows:

$$\begin{aligned} \sigma_{zz} &= f_1 e^{i(-bx + \frac{\omega t^\alpha}{\alpha})}, \\ \sigma_{xz} &= 0, \quad \Theta = f_2 e^{i(-bx + \frac{\omega t^\alpha}{\alpha})}, \\ \frac{\partial \psi}{\partial z} &= 0, \end{aligned} \tag{38}$$

where f_1 and f_2 are constants.

Mechanical boundary condition: When a mechanical force acts on the surface of the half-space, the boundary condition $\sigma_{zz} = f_1 e^{i(-bx + \frac{\omega t^\alpha}{\alpha})}$ is applied. When the surface of the half-space is free of traction, the boundary condition $\sigma_{xz} = 0$ is used. $\Theta = f_2 e^{i(-bx + \frac{\omega t^\alpha}{\alpha})}$ indicates that The temperature at the boundary is harmonically varying in space and time, with amplitude And the fourth boundary condition, $\frac{\partial \psi}{\partial z} = 0$, is a constraint on the rate of change of microelongation normal to the surface.

Substituting the assumed functional forms of the variables into the boundary conditions given in (38), yields a set of relations that the system parameters must satisfy. This procedure leads to a system of four linear equations. To evaluate the unknown constants (G_1, G_2, G_3, G_4) the system is solved using the inverse matrix technique as shown below:

$$\begin{bmatrix} G_1 \\ G_2 \\ G_3 \\ G_4 \end{bmatrix} = \begin{bmatrix} \vartheta_{51} & \vartheta_{52} & \vartheta_{53} & \vartheta_{54} \\ \vartheta_{61} & \vartheta_{62} & \vartheta_{63} & \vartheta_{64} \\ \vartheta_{21} & \vartheta_{22} & \vartheta_{23} & \vartheta_{24} \\ -r_1 \vartheta_{31} & -r_2 \vartheta_{32} & -r_3 \vartheta_{33} & -r_4 \vartheta_{34} \end{bmatrix}^{-1} \begin{bmatrix} f_1 \\ 0 \\ f_2 \\ 0 \end{bmatrix} \tag{39}$$

5. Numerical discussions and results

This section provides a set of graphical plots that illustrate the solutions in two-dimensional form. These visual results are essential for demonstrating the evolution of the various field variables, thereby deepening the understanding of their physical behavior and the qualitative trends inherent in the solutions. For numerical evaluation, aluminum–epoxy is chosen as the thermo-elastic material due to its well-documented thermal and mechanical properties, which make it a suitable and reliable choice for investigating thermo-elastic interactions. The material constants required for the analysis

are taken from reputable references, and the numerical values used in the computations are summarized in the following [47]

$$\begin{aligned} j^* &= 0.196 \times 10^{-4} \text{ m}^2, & \lambda &= 7.59 \times 10^{10} \frac{\text{N}}{\text{m}^2}, \\ \rho &= 2190 \frac{\text{kg}}{\text{m}^3}, & \mu &= 1.89 \times 10^{10} \frac{\text{N}}{\text{m}^2}, \\ \gamma_0 &= 0.50 \times 10^5 \frac{\text{N}}{\text{m}^2 \cdot \text{K}}, & \lambda_0 &= 0.37 \times 10^{10} \frac{\text{N}}{\text{m}^2}, \\ \gamma_1 &= 0.50 \times 10^5 \frac{\text{N}}{\text{m}^2 \cdot \text{K}}, & c_e &= 966 \frac{\text{J}}{\text{kg} \cdot \text{K}}, \\ k &= 252 \frac{\text{J}}{\text{m} \cdot \text{s} \cdot \text{K}}, & T_0 &= 293^\circ \text{ K}, \\ \lambda_1 &= 0.37 \times 10^{10} \frac{\text{N}}{\text{m}^2}, & a_0 &= 0.61 \times 10^{-10} \frac{\text{N}}{\text{m}^2}, \\ \tau_\theta &= 0.6 \text{ s}, & \tau_q &= 0.9 \text{ s}, & f_1 &= 15, & f_2 &= 10. \end{aligned}$$

In this paper, the numerical evaluations of all physical quantities are performed at the nondimensional time $t = 0.2$ across the spatial domain $0 \leq z \leq 2$ and along the surface $x = 1.5$. The numerical technique employed in this work is used to examine and describe the behavior and variations of the physical quantities u , w , ψ , σ_{zz} , and σ_{xz} . The graphical representations illustrate the theoretical predictions based on the L-S theory, DPL model, and RDPL model providing a comparative understanding of their respective behaviors. Figure 1, Figure 2, Figure 3, Figure 4 and Figure 5 present a detailed comparative analysis of the L-S theory, DPL model, and RDPL model, highlighting the effect of rotation for a fractional-order parameter of $\alpha = 0.5$. Figure 1 illustrates the influence of rotation on the horizontal displacement component u for the L-S theory, DPL model, and RDPL model. The results reveal that rotational motion induces noticeable modifications in the displacement amplitude due to the combined action of centrifugal and Coriolis forces. A reduction in the magnitude of u is observed as rotation increases, with the RDPL model exhibiting the strongest damping behavior owing to its enhanced phase-lag structure. Figure 2 presents the impact of rotation on the vertical displacement component w . The results indicate that rotation alters the vertical wave profile by introducing additional inertial coupling. The displacement amplitude decreases with increasing rotation, highlighting the stabilizing effect of rotational forces. Among the models, the RDPL model produces the most attenuated response, whereas the L-S theory yields a comparatively higher amplitude. Figure 3 depicts the distribution of the micro-elongation field ψ under different rotational conditions. The figure illustrates that rotation significantly microstructural deformation within the material. The reduction in ψ amplitude clarifies that the effect of rotational inertia on internal micro-elongated phenomena, with the RDPL model capturing this behavior most effectively. Figure 4 clarifies the impact of rotation on the normal stress component σ_{zz} . It is noted that rotational motion alters the stress distribution significantly, increasing compressive and tensile variations. RDPL model predicts

lower stress amplitudes, showing better damping, while L-S theory produces the highest wave amplitude due to weaker relaxation effects. Figure 5 shows the influence of rotation on tangential stress σ_{xz} . The tangential stress becomes more oscillatory and shifts in amplitude depending on the thermo-elastic model. RDPL model again displays the strongest attenuation, confirming its suitability for rotating systems. Figure 1 shows the variation of the field quantity under consideration with respect to the spatial coordinate, illustrating the influence of rotation and the fractional conformable derivative within the RDPL thermoelastic framework. The results demonstrate that rotation significantly modifies the wave amplitude and attenuation, while the fractional parameter controls the memory-dependent behavior of the micro-elongated thermoelastic response. The obtained results are compared with previously published studies. It is observed that the numerical trends illustrated in Figure 1, Figure 2, Figure 3, Figure 4, Figure 5 and Figure 6 are in close qualitative agreement with those reported in [41, 48] when the current model is reduced to their special cases by neglecting RDPL model and fractional-order effects. The remaining differences are attributed to the combined influence of rotation, micro-elongated behavior, and the fractional conformable RDPL model, demonstrating the extended capability and novelty of the present analysis. Figure 7, Figure 8, Figure 9, Figure 10, Figure 11 and Figure 12 are introduced to clarify and interpret the variation of the previously discussed physical quantities considering the effect of rotation ($\Omega = 2$) within the RDPL model. These graphical results present an in-depth depiction of how the quantities respond for three different values of the fractional order parameter, denoted by α . Figure 7 shows the effect of different fractional orders α on the horizontal displacement u . Increasing the fractional order enhances effects in heat conduction, which slows thermal diffusion and modifies the displacement amplitude. Lower α produces higher oscillations, indicating stronger relaxation effects. Figure 8 displays the influence of varying the fractional order α on the vertical displacement w . The results show that lower fractional orders intensify heat conduction, generating sharper gradients and larger oscillations in the displacement field. Conversely, higher α values lead to a more diffusive thermal response and reduced mechanical disturbance. Figure 9 depicts the effect of different fractional orders α on the micro-elongation field ψ . A clear increase in micro-elongation amplitude is observed for smaller α , indicating that fractional thermal behavior enhances internal micro-elongation effects. Higher fractional orders yield smoother and more attenuated, reflecting faster thermal relaxation. Figure 10 demonstrates the distribution of the normal stress σ_{zz} for several fractional orders α . The results show that decreasing α produces stronger stress concentrations and sharper oscillations due to intensified thermal lag. In contrast, higher fractional orders lead to more uniform stress distributions with reduced peak values. Figure 11 presents the ef-

fect of the fractional-order parameter α on the tangential stress σ_{xz} . Lower values of α amplify the tangential response and increase oscillatory behavior, highlighting the thermal-mechanical coupling. As α increases, the tangential stresses become smoother and exhibit reduced amplitude. Figure 12 shows the effect of varying the fractional order α on the temperature Θ . The results show that lower fractional orders intensify heat conduction, generating sharper gradients and larger oscillations in the temperature. Conversely, higher α values lead to a more diffusive thermal response.

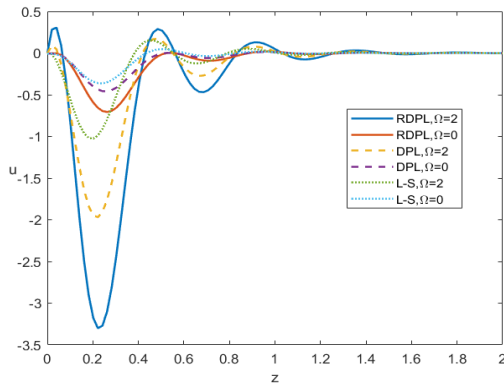


Figure 1. Impact of rotation on u .

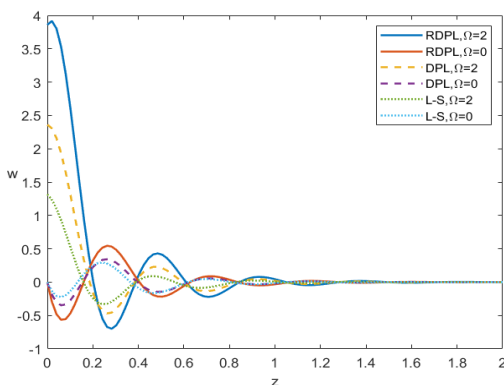


Figure 2. Impact of rotation on w .

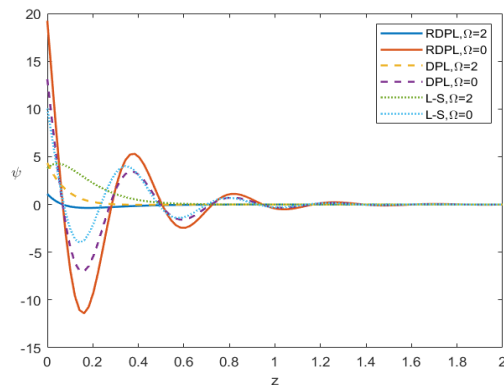


Figure 3. Impact of rotation on ψ .

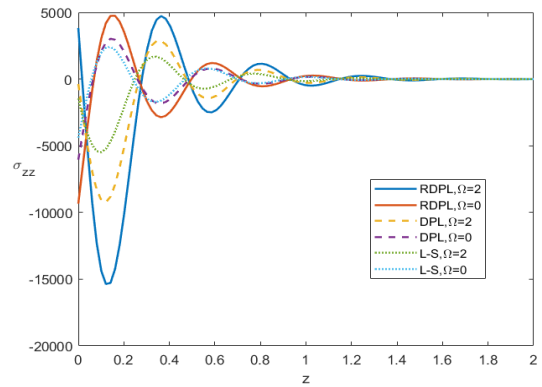


Figure 4. Impact of rotation on σ_{zz} .

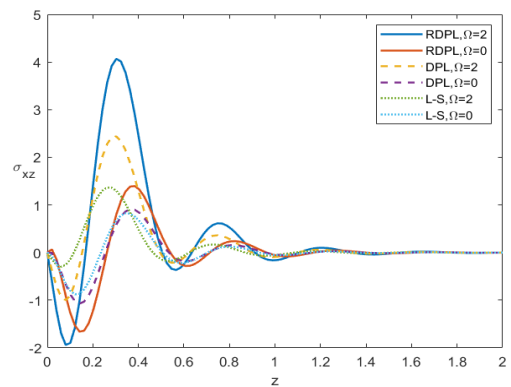


Figure 5. Impact of rotation on σ_{xz} .

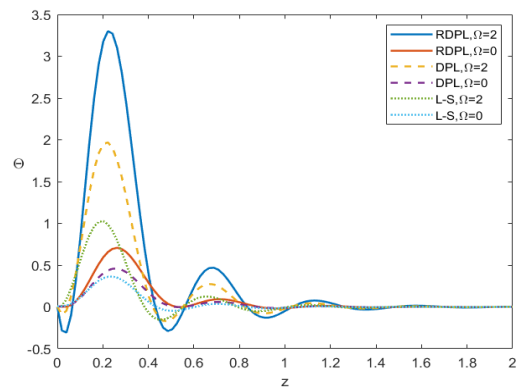


Figure 6. Impact of rotation on Θ .

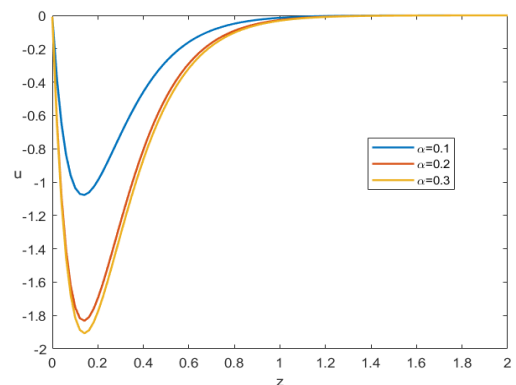


Figure 7. Effect of different fractional order (α) on u .

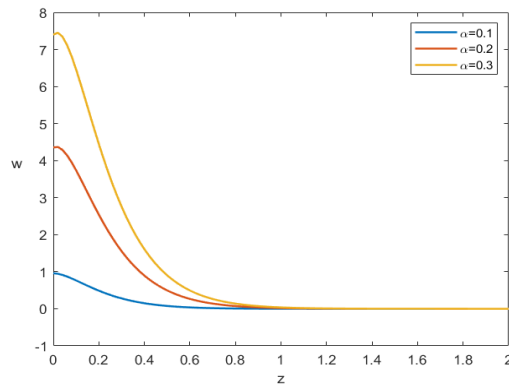


Figure 8. Effect of different fractional order (α) on w .

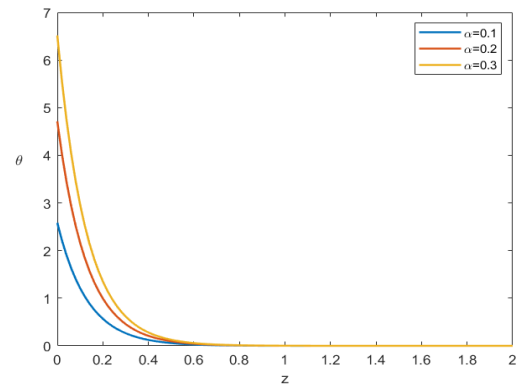


Figure 12. Effect of different fractional order (α) on θ .

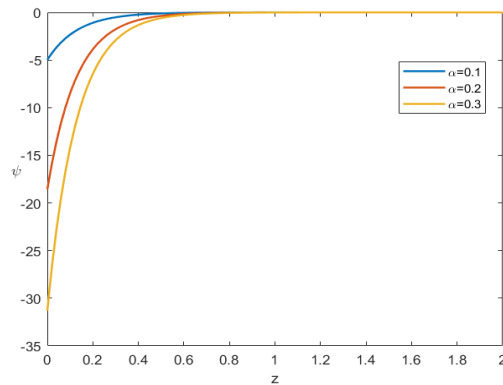


Figure 9. Effect of different fractional order (α) on ψ .

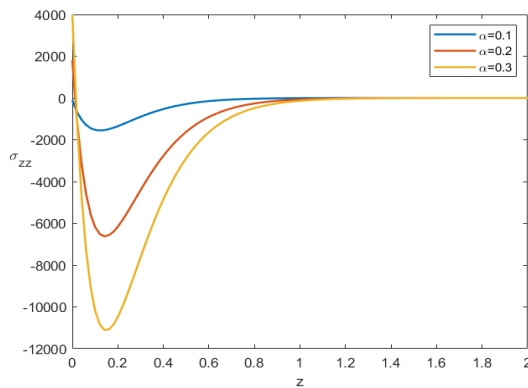


Figure 10. Effect of different fractional order (α) on σ_{zz} .

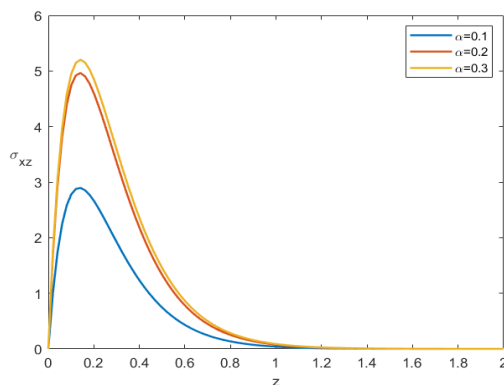


Figure 11. Effect of different fractional order (α) on σ_{xz} .

6. Conclusion

In this paper, the thermo-elastic response of a rotating micro-elongated half-space has been thoroughly investigated within the frameworks of the L-S theory, the DPL model, and the RDPL model, incorporating the effect of fractional conformable derivatives. The governing equations describing displacement, micro-elongation, and thermal fields were formulated, non-dimensionalized, and solved analytically using a normal mode approach. This allowed explicit solutions for all primary field variables to be obtained, facilitating a comprehensive investigation of wave propagation under rotation. The present results provide important insights into the behavior of rotating microstructured materials subjected to thermal loading. Such materials are commonly encountered in advanced composite components used in turbines, rotating machinery, and aerospace structures, where rotation-induced effects significantly influence thermo-elastic wave propagation and heat transfer characteristics. Rotation introduces additional inertial effects primarily the Coriolis and centrifugal forces which alter wave amplitude, propagation speed, and attenuation characteristics. These effects are particularly evident in the displacement fields, where both horizontal and vertical components experience noticeable modulation in response to increasing rotational speed. The micro-elongation field ψ is likewise strongly influenced by rotation, with its amplitude decreasing as rotational effects intensify, reflecting the inhibitory action of inertial forces on microstructural deformation. The stress components σ_{zz} and σ_{xz} exhibit distinct changes in magnitude and spatial distribution, illustrating how rotation enhances tangential stress interaction and alters normal stress oscillations within the medium. A comparative assessment among L-S theory, DPL model, and RDPL model reveals that the RDPL model consistently provides a more attenuated and physically realistic response due to its refined phase-lag formulation. This model proves especially effective in capturing the combined effects of fractional heat conduction, micro-elongation, and rotation.

The present analysis can be extended in several important directions. Future studies may consider two- and

three-dimensional configurations and more realistic geometries to better represent practical engineering structures. The influence of variable rotational speed, initial stress, magnetic fields, or poroplastic effects may also be incorporated to further enrich the physical model. In addition, comparing the fractional conformable derivative with other fractional operators could provide deeper insight into thermo-elastic behavior.

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Authors contributions

All the authors have participated sufficiently in the intellectual content, conception and design of this work or the analysis and interpretation of the data (when applicable), as well as the writing of the manuscript.

Availability of data and materials

The datasets used and/or analyzed during the current study are available from the corresponding author upon reasonable request.

Conflict of interests

The authors declare that they have no competing interests.

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